<table>
<thead>
<tr>
<th>AIRPLANE MODEL</th>
<th>T.C. NO.</th>
<th>DESCRIPTION OF TYPE DESIGN CHANGE</th>
<th>APPROVED DATE</th>
</tr>
</thead>
<tbody>
<tr>
<td>Gulfstream American Corp. AA-5B</td>
<td>A16EA</td>
<td>Install BA-120 filter assembly as directed by AC drawing No. 8994656</td>
<td>July 20, 1979</td>
</tr>
<tr>
<td>GA-7</td>
<td>A17S0</td>
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<td>July 20, 1979</td>
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</thead>
<tbody>
<tr>
<td>PA-28-151 PA-28-161</td>
<td>2A13</td>
<td>Install according to original Manufacturers Service Instructions.</td>
<td>June 1, 1973</td>
</tr>
</tbody>
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<tr>
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</thead>
<tbody>
<tr>
<td>Gulfstream American AA-5A</td>
<td>A16EA</td>
<td>Install according to original Manufacturers Service Instructions.</td>
<td>May 15, 1974</td>
</tr>
</tbody>
</table>
United States Of America
Department of Transportation – Federal Aviation Administration

Supplemental Type Certificate

Number SA693CE

This Certificate issued to Brackett Aero Filters, Inc.
7052 Government Way
Kingman, Arizona 86401

certifies that the change in the type design for the following product with the limitations and
conditions therefore as specified hereon meets the airworthiness requirements of Part Specified
under: “Certification Basis” on the applicable Data Sheets or Aircraft Specifications
Regulations.

Original Product Type Certificate Number: As shown on Approved Model List(s)

Make: “
Model: “

Note: FAA – Approved Model List(s) form a part of this Certificate.

Description of Type Design Change: Replace existing engine induction air filter with AC Spark Plug
Division filter in accordance with Approved AC Filter List(s).

Limitations and Conditions: This approval should not be extended to other specific airplanes of these
models on which previously approved modifications are incorporated unless it is determined that the
interrelationship between this change and any of those other previously approved modifications will
introduce no adverse effect upon the airworthiness of the airplanes. If the holder agrees to permit another
person to use this certificate to alter the product, the holder shall give the other person written evidence of
that permission.

This certificate and the supporting data which is the basis for approval shall remain in effect
until surrendered, suspended, revoked or a termination date is otherwise established by the
Administrator of the Federal Aviation Administration.

Date of application: March 2, 1970

Date of issuance: March 26, 1970


Date amended:

By direction of the Administrator

Michael A. Murphy
(Signature)
Manager, Tech. and Admin. Support Staff
Los Angeles Aircraft Certification Office

Any alteration of this certificate is punishable by a fine of not exceeding $1,000, or imprisonment not exceeding 3 years, or
both.

FAA Form 8110-2(10-68) Page 1 of 2 This certificate may be transferred in accordance with FAR 21.47.
INSPECTION INTERVALS: Pre-flight inspections, engine backfire inspection, 100 hour inspections, annual inspections, filter element replacements.

INSPECTION PROCEDURES

A. Pre-flight inspection: Per Pilots Operating Handbook, check filter assembly for security, damage or 50% contamination of element face. If found report to maintenance personnel prior to flight.

B. Engine start-up backfire inspection: Prior to flight, check the entire intake system for security or damage. If a fire was present, the downstream face of the foam element will show erosion. If any irregularities are found see Chart I and also refer to the Aircraft Maintenance Manual for the intake system.

C. At element replacement intervals: With the element removed, inspect the filter grill, filter frame, filter mountings and entire intake system for security, wear and any deformation. Note: On filter assemblies with gaskets, visually inspect inside and outside of frame for any signs of gasket looseness, movement or deterioration. If found refer to Chart I or the proper maintenance manual for your aircraft or component.

<table>
<thead>
<tr>
<th>PART</th>
<th>REPLACE</th>
<th>REROWK</th>
<th>INSTRUCTIONS</th>
</tr>
</thead>
<tbody>
<tr>
<td>FRAME</td>
<td>X</td>
<td></td>
<td>COMPLETE NEW ASSEMBLY</td>
</tr>
<tr>
<td>GASKET</td>
<td>X</td>
<td></td>
<td>ON FILTER FRAME, REMOVE OLD NEOPRENE GASKET AND ALL TRACES OF ADHESIVE DOWN TO A CLEAN ANODIZED FRAME SURFACE. USE ADHESIVE 3M#947 OR DOW CORNING RTV-732. COAT ENTIRE MATING SURFACE (GASKET TO FRAME). APPLY ADHESIVE FOLLOWING MANUFACTURER'S LABEL DIRECTIONS. WHEN FRAME AND GASKET ARE PLACED TOGETHER, CLAMP OR WEIGHT DOWN AT .75 LB./SQ. IN. OF CONTACT AREA. ALLOW TO CURE 24 HOURS PRIOR TO INSTALLATION. PRIOR TO REINSTALLATION OF FILTER, CHECK AIRBOX MATING SURFACE FOR IRREGULARITIES. IF FOUND, CORRECT PER MANUFACTURER'S REQUIREMENTS. UPON REINSTALLATION CHECK THAT THE GASKET MAKES 100% CONTACT. THE GASKET SHOULD BE COMPRESSED 50% FOR OPTIMUM SEAL.</td>
</tr>
<tr>
<td>GRILL</td>
<td>X</td>
<td></td>
<td>REPLACE WITH NEW GRILL</td>
</tr>
<tr>
<td>HARDWARE</td>
<td>X</td>
<td></td>
<td>REPLACE UNSERVICEABLE WITH NEW</td>
</tr>
<tr>
<td>SCREEN</td>
<td>X</td>
<td></td>
<td>COMPLETE NEW ASSEMBLY</td>
</tr>
<tr>
<td>SCREEN/GASKET</td>
<td>X</td>
<td></td>
<td>COMPLETE NEW ASSEMBLY</td>
</tr>
<tr>
<td>ELEMENT</td>
<td>X</td>
<td></td>
<td>REPLACE ELEMENT</td>
</tr>
</tbody>
</table>

DATE: 3-16-94

CHART 1